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Loss of Mars Polar Lander December 3, 1999 in the vicinity of space Mars

Masayuki Nakao (Institute of Engineering Innovation, School of Engineering, The University of Tokyo)

NASA (U.S. National Aer onautics and Space Administration) lost contact with its spacecraft "Mars P olar Lander (MPL)" just prior to it s scheduled atmospheric entry and landing on the south polar region of Mars. Communication was never regained. It is assumed that the MPL's legs contact sensor tripped during descent, which signaled false indication that the spacecraft had landed, resulting in a premature shutdown of the engines 40 meters ab ove the s urface. MPL cr ashed on the Martian s urface, and i ts two Deep S pace 2 Mi croprobes (DS2) were presumably destroyed upon impact.



Figure 1. Artist's Concept of Mars Polar Lander [5]

1. Event

NASA lost contact with the spacecraft "Mars Polar Lander (MPL)" just prior to its scheduled atmospheric entry and landing in the so uth-polar region of Mars. The P olar Lander was carrying two De ep S pace 2 (DS2) Microprobes designed to study the Martian weather, climate, and water and carbon dioxide levels. It is not k nown whether the lander r eached the t erminal descent propulsion phase. The microprobes were never located.

2. Course

NASA launched MPL on January 3, 1999.

After a 11-month travel (7.57 million km) to Mars, MPL was about to enter the Mars' atmosphere in the early morning of December 3. To land on the so uth polar region of Mars, a f inal trajectory-correction maneuver was executed 6.5 hours before entry.

After its initial high-speed entry into the Martian atmosphere, the MPL's descent plan was as follows.

(Figure 2 illustrates the entry, descent and landing phase of MPL.)

At about an altitude of 7.5 km to landing, while the Lander was still moving at 250 km/sec, it would deploy its 8.4-meter-wide parachute. Although the par achute would substantially slow the d escent, it would be unable to slow it below 80 m/sec in Mars' thin at mosphere. 10 seconds after parachute deployment, the heat shield would be released to cr ash onto the surface s o that the friction produced by the Martian atmosphere would quickly decelerate the vessel from its initial velocity. 16 seconds after the release of the heat shield, the MPL's three legs, folded up to fit inside the shield, would swing down and latch in position. When the Lander was about 1800 meters above the surface and descending at 80 m/sec, the parachute and backshell would separate from MPL, and in 0.5 seconds, the three clusters of landing thrusters would start firing. The engine thrust would be modulated during the remaining descent in accord with data from the probe's multi-beam landing radar to gradually slow the vessel to a gentle 2.4 m/sec final descent.



Figure 2. MPL's Entry, Descent and Landing Phase [1]

At 12:02 pm PST (U.S. Pacific Standard Time), MPL entered the atmosphere. The high temperature from the heat during MPL's descent in the atmosphere closed off the communication signal with the Earth. MPL touchdown was expected at 12:14 pm PST, with a data transmission to Earth scheduled to begin 24 minutes later. However, no communication from MPL was received.

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After detaching from MPL, the two DS2 probes were to extend and drill deeper into the ground to search for water as shown in Figure 3. The first data from the DS2 probes was expected to reach the Earth at 7:25 pm PS T on December 4, ab out 7 hours after MPL touchdown. How ever, no communication from the probes was received.



Figure 3. DS2 Entry, Descent, and Impact Sequence with Potential Failure Modes [2]

3. Cause

- (1) Accidental tripping of the MPL's landing legs contact sensors during descent NASA concluded that the MPL's onboard systems confused the jolt from the deployment of a landing leg with ground contact, and s hutdown the engines prematurely at 40 meters a ltitude d own to the actual landing. MPL was assumed to have crashed on the Martian surface at 22 m/sec (about 80 km/h).
- (2) A design flaw in the ground-contact switch system When the MPL's landing legs first swung down to lock into position after the release of the heat shield, sufficient for ce on fl exible parts of the legs bounced the leg s upw ards again, and t riggered the ground-contact switch to send false indicator signals to the computer system.

The Failure Review Board for the Mars Polar Lander discovered that this event repeated consistently in lab tests after the incident. Such behavior, however, was nev er detected during MPL's prel aunch tests, because different teams conducted MPL's tests before and after the legs' fold-down procedure.

(3) While the above cause was the most likely scenario, sever al other events were still possible. MPL might have burnt out in the Martian atmosphere, or it might have failed to land properly on the rugged Martian terrain.

4. Immediate Action

NASA first assumed that the communication antenna was not pointed towards the earth, or a minor trouble caused the computer to e nter a te mporary "sleep" mode and tried to send signals to change the antenna angles or wait for the computer to "wake."

NASA also used the Mars Global Surveyor satellite orbiting Mars in an attempt to communicate with MPL, however, all attempts failed. On January 17, 2000, NA SA ended the intermittent search for MPL and the DS2 microprobes.

NASA appointed Thomas Young, a previous executive vice president of Lockheed Martin Corp., to chair the Mars P rogram I ndependent Assessment Team comprised of 18 a erospace s cientists and review the failure.

NASA canceled its 2001 Mars Surveyor Program to launch a lander with landing gears similar to the one on the lost MPL.

5. Countermeasure

Unknown.

6. Summary

The cost of s pace development is enormous and efforts to cut the cost are necessary. NAS A, however, should have clarified where it can cut the cost or not (at the technology level at the time), otherwise, cost reduction can directly link to a failure and all the expenses can go down the drain.

The sp ecial antenna for communication during descent was taken of ft o cut the cost and MPL's communication with earth was never reestablished after it was cut of f12 min utes prior tol anding. Accordingly, there were no data to tell if the planned sequence of processes were carried out and where in the course a malfunction happened.

The MPIAT report findings included the following:

- The 1998 Mars Surveyor Program, which comprised launches of the Mars Climate Orbiter and the Mars Polar Lander, was challenged by inadequate budget cost-caps and was significantly underfunded by 30%.
- The DS2 development had competent, but inexperienced, project managers.
- Senior m anagement with y ears of experience at NA SA Headquarters, JP L, and Loc kheed Mar tin Astronautics failed to have the follower managers acquire proper management skills.
- The lack of device for communication during the entry, descent, and landing data for MPL was a major design mistake.
- The DS2 microprobes had unprofessional designs and were put through an inadequate test program; they were not at all ready for the mission.
- The program was significantly understaffed. The development team of government and civil workers overworked, sometimes 60 to 80 hours a week.

7. Knowledge

- (1) Modifications to cut cost have the danger of turning into a cause of failure
 - To achieve a soft landing, Lunar Surveyors and Viking were equipped with a radar altimeter to control the final approach to the landing site. Terminal descent engines were designed to shutdown at 3 meters on the r adar altimeter readings. The MP L landing system was simplified so the terminal d escent engines would shutdown as soon as the on board systems received a ground-contact signal from a landing leg.
- (2) Multiple team operations have pitfalls in the boundary area among teams.
- (3) Adequate funding, staffing and scheduling are critical to successful program implementation. Such co st-cutting sometime s su cceed, howe ver, blindly believing in t hem sh ould be avoided. A successful experience in cost-cutting can encourage skipping necessary processes and lead to failure.
- (4) Giving low priority in transfer of technologies and management skills lead to failures.

8. Background

NASA Administrator Daniel Goldin implemented the Faster, Better, Cheaper (FBC) philosophy in 1992 in response to budget reductions. He stopped repeating 10s of years projects with huge budgets and started to encourage subletting tasks to c ontractors and c ivil-service workforce so that the agen cy could shorten development time, reduce cost, and increase the scientific return by flying more missions in less time. Developed in accord with this FBC approach, the Mars Pathfinder lander and rover successfully reached the rocky terrain of the Martian equator on July 1997. Two months later, the camera-equipped Mars Global

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Surveyor orbiter successfully reached Mars and continues to send images of unprecedented detail to the Earth.

However, the FBC effort was undermined by inadequate funding. The 1998 Mars Surveyor Program only had about half of the successf ul Mar s Pat hfinder lander and Mar s Global S urveyor or biter miss ions, significantly underfunded by a t lea st 3 0%. The numb er of r esident JPL (Jet Prop ulsion Lab oratory i n Pasadena, California) staff was reduced to 15. One third of previous staff was allocated to the flight control team of three Mars Surveyor Programs (Mars Global Surveyor, Mars Climate Orbiter a nd Mars Polar Lander). Preflight testing was inadequate. The loss of Mars Polar Lan der followed the mission failure of Mars Climate Orbiter, which was launched on December 11, 1998 and lost in deep space due to suspected navigational errors on September 23, 1999.

References

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